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### **Stability of Hypersonic Boundary Layer on Porous Wall with Regular Microstructure**

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Speaker Could Not Present due to Visa Problems. Paper is Available. Here, a Brief Summary by the Session Chair

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## Caltech<sup>\*</sup> experiments confirm the theoretical concept<sup>†</sup> of hypersonic laminar flow control



Sharp cone model with porous surface

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5x10<sup>6</sup>

4x10

3x10

2x10

1x10<sup>6</sup>

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3

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Perforated sheet with blind cylindrical holes:

- hole diameter 60 microns
- hole depth 500 microns
- spacing 100 microns





1 mm

Summary of Caltech data (*H*<sub>0</sub> is total enthalpy)

7

black squares - solid wall

open circles - porous wall

8

H<sub>a</sub>, MJ/kg

arrowed circles - flow is totally laminar

9

11

12

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10

<sup>\*</sup>Rasheed, A., Hornung, H.G., Fedorov, A.V., and Malmuth, N.D., *AIAA J.*, 40, No. 3, 2002. †Fedorov & Malmuth et al., *AIAA J.*, 39, No. 4, 2001

## ITAM stability experiments on felt-metal coating agree well with LST\*



\*Fedorov A., Shiplyuk, A., Maslov, A., Burov, E., and Malmuth, N., AIAA Paper No. 2003-1270, 2003; *JFM*, Vol. 479, 2003, pp. 99-124.

## **Objectives**

- Perform theoretical and experimental studies of porous coating of regular microstructure
- Include gas rarefaction effect in theoretical model of acoustic absorption
- Conduct stability measurements of natural and artificially excited disturbances
- Compare LST predictions with experiment

# Porous coating is similar to UAC of Caltech experiments<sup>\*</sup>







Stainless steel sheet perforated with equally spaced cylindrical holes:

- Diameter 50  $\mu\text{m}$  on face side
- Spacing 100 μm
- Depth 450 μm
- Porosity 20%

<sup>\*</sup>Rasheed, A., Hornung, H.G., Fedorov, A.V., and Malmuth, N.D., AIAA J., 40, No. 3, 2002.

# Sharp-cone model tested in the T-326 wind tunnel of ITAM





### Perforated sheet was tensed onto the model to avoid cavities

Base part with the porous coating

#### Spectra of Natural Disturbances on Solid and Porous Sides (M=5.95, T0=390K, Tw/T0=0.80-0.84)



#### Artificially Excited Wave Packets in Boundary Layer on Porous Walls

![](_page_7_Figure_1.jpeg)

frequency 275 kHz

![](_page_7_Figure_3.jpeg)

#### Amplitude vs. circumferential angle

 $\beta$ -spectra

## Dominant component of artificially excited wave packet is two-dimensional wave of $\beta$ =0

#### Phase Speeds and Amplitudes of Natural and Artificial Disturbances

![](_page_8_Figure_1.jpeg)

Natural disturbances of high frequency are predominantly 2-D waves of the second mode.

# For Kn<1, gas rarefaction increases performance of porous coating

![](_page_9_Figure_1.jpeg)

Maximum (versus frequency) growth rate of second mode

## Amplification of Artificial and Natural Disturbances of 275kHz (Experiment and LST)

![](_page_10_Figure_1.jpeg)

Theory agrees well with experiment on both porous and solid walls

### Conclusions

- Experimental and theoretical studies of hypersonic boundary layer stability were performed for ultrasonically absorptive coatings (UAC) of regular microstructure
- Under natural conditions, UAC stabilizes the second mode and weakly affects the first mode that is consistent with theoretical predictions
- Natural disturbances of high-frequency band are predominantly 2-D waves of the second mode
- UAC stabilizes high-frequency wave packets of second-mode instability
- Theoretical predictions of second-mode growth agree well with experiment for both solid and porous walls
- For Kn<1, gas rarefaction increases UAC performance