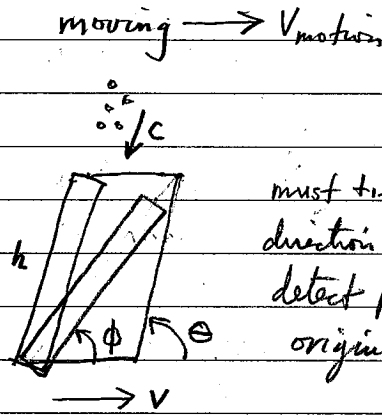
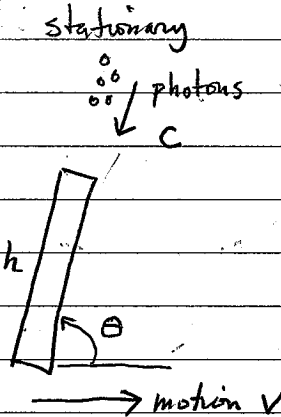
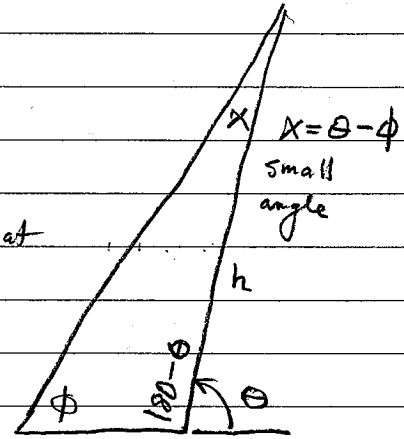


if you move you must tilt the umbrella in the direction of motion to use umbrella as a sensor to detect direction of rain motion, you must tilt it back.

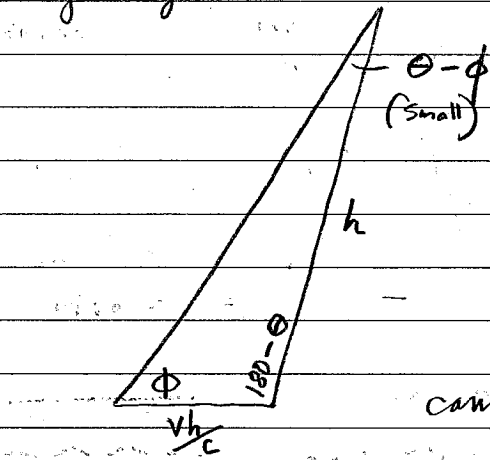


must tilt telescope in direction of motion to detect photons coming at original angle θ



$d = vt$, $t = d/v$, $c = 3 \times 10^8 \text{ m/s}$
 photon enters tube (length = h), takes $t = h/c$ sec. to reach the detector; telescope moving at speed $= V$ travels distance: $v \cdot t = V \cdot h/c$. label triangle edges:

$180 - \theta + \phi + x = 180$
 $x = \theta - \phi$ (small angle)



photon traverses tube while telescope moves $v h/c$ distance.

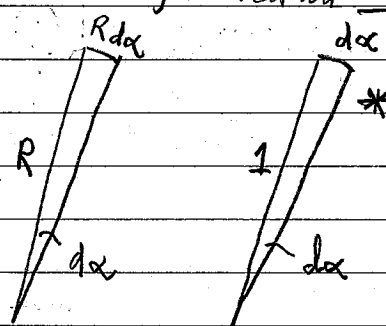
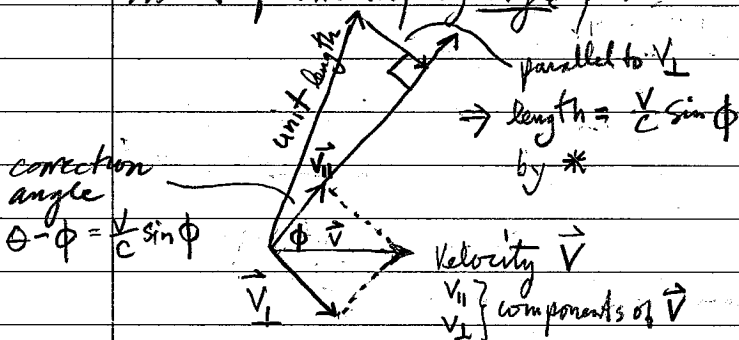
law of sines: $\frac{v h/c}{\sin(\theta - \phi)} = \frac{h}{\sin \phi}$

small angle $\theta - \phi$: $\frac{v h/c}{\theta - \phi} = \frac{h}{\sin \phi}$

cancel h : $\frac{v/c}{\theta - \phi} = \frac{1}{\sin \phi}$

$\theta - \phi = \frac{v}{c} \sin \phi$ correction angle to tilt telescope.

instead of correcting by angle, we correct view vector by correction vector



$$\left. \begin{aligned} \text{correction angle} &= \frac{v}{c} \sin \phi = \frac{|\vec{v}|}{c} \sin \phi \\ |\vec{v}_{\perp}| &= |\vec{v}| \sin \phi \end{aligned} \right\} \text{angle} = \frac{|\vec{v}_{\perp}|}{c} \Rightarrow \text{arc length} = \frac{|\vec{v}_{\perp}|}{c}$$

 $(R=1)$
 (in direction \vec{v}_{\perp})

\Rightarrow vector correction = $\frac{\vec{v}_{\perp}}{c}$ (our algorithm deals with relative velocities not absolute velocities)

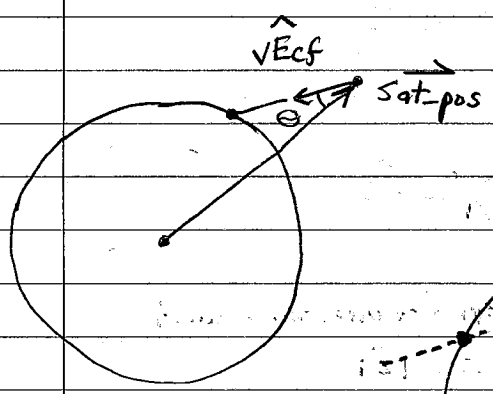
5
$$\vec{v}_{Ecf}^{Corr} = \hat{v}_{Ecf} - \frac{\vec{v}_{Rel}_{\perp}}{c}$$

↑
 corrected view vector

↑
 original view vector

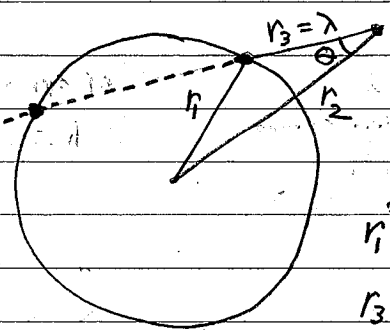
correction vector negative since we undo the tilt of the telescope.

$\hat{\square}$: implies unit vector



1
$$\cos \theta = \hat{v}_{Ecf} \cdot \frac{-\vec{Sat_pos}}{|\vec{Sat_pos}|}$$

 $= \hat{v}_{Ecf} \cdot -\hat{Sat_pos}$



use law of cosines for the triangle

$r_1^2 = r_2^2 + r_3^2 - 2r_2r_3 \cos \theta$

$r_3 = \lambda$, $r_1 = \text{mean radius of earth}$
 $r_1 = r_e = 6371 \text{ km}$

$r_e^2 = r_2^2 + \lambda^2 - 2r_2\lambda \cos \theta$
 $r_2^2 + \lambda^2 - r_e^2 - 2r_2\lambda \cos \theta = 0$

use quadratic formula to solve for λ
 $A\lambda^2 + B\lambda + C = 0$, $\lambda = \frac{-B \pm \sqrt{B^2 - 4AC}}{2A}$

{ choose minus since we want smaller solution for λ }

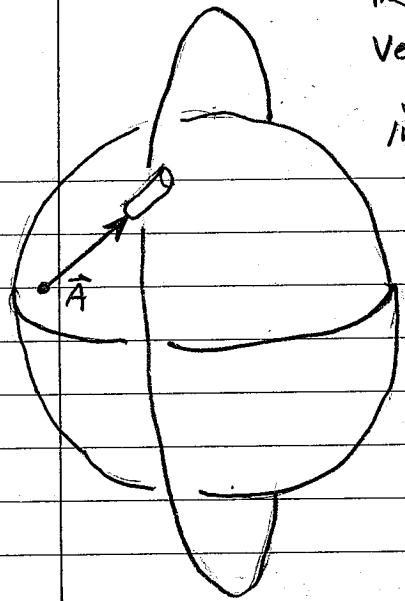
$A=1, B=-2r_2 \cos \theta, C=r_2^2 - r_e^2, r_2 = |\vec{Sat_pos}|$

$$\lambda = \frac{2r_2 \cos \theta - \sqrt{4r_2^2 \cos^2 \theta - 4r_2^2 + 4r_e^2}}{2}$$

$$\lambda = r_2 \cos \theta - \sqrt{r_2^2 \cos^2 \theta + r_e^2 - r_2^2}^{1/2}$$

2
$$\lambda = |\vec{Sat_pos}| \cos \theta - \left[|\vec{Sat_pos}|^2 \cos^2 \theta + r_e^2 - |\vec{Sat_pos}|^2 \right]^{1/2}$$

The following relation works for \vec{A} either position or velocity, with $\frac{d\vec{A}}{dt}$ respectively velocity or acceleration



in our case \vec{A} is position, $\frac{d\vec{A}}{dt}$ is velocity
the equation relates fixed and rotating reference frames.

$$\frac{d\vec{A}}{dt}_{\text{fixed}} = \frac{d\vec{A}}{dt}_{\text{rotating}} + \vec{\omega} \times \vec{A}$$

mean sidereal earth rotation rate vector

\vec{A} : relative position
 $\frac{d\vec{A}}{dt}$: relative velocity

$$\vec{\omega} = \begin{bmatrix} 0 \\ 0 \\ 7.292115833 \times 10^{-5} \end{bmatrix} \text{ Rad/sec}$$

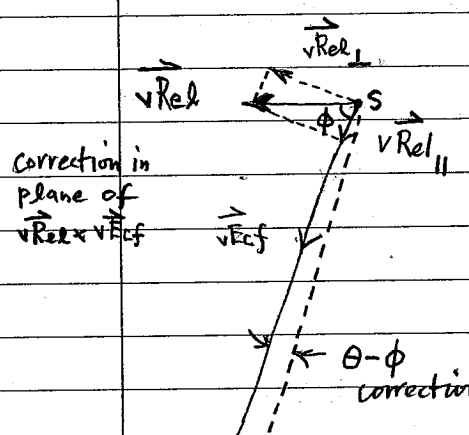
$$\frac{d\vec{A}}{dt}_{\text{fixed}} = \begin{bmatrix} v_x \\ v_y \\ v_z \end{bmatrix} + \vec{\omega} \times \lambda(-\hat{v}_{Ecf})$$

relative velocity fixed

interpolated velocity vector from ephemeris (rotating)

\hat{v}_{Ecf} opposite direction from \vec{A}
 λ from [2]

$$\boxed{3} \quad \vec{Sat vel rel} = \begin{bmatrix} v_x \\ v_y \\ v_z \end{bmatrix} - \lambda(\vec{\omega} \times \hat{v}_{Ecf})$$



$$\begin{aligned} \vec{v}_{Rel} &= \vec{v}_{Rel_{||}} + \vec{v}_{Rel_{\perp}} \\ \vec{v}_{Rel_{\perp}} &= \vec{v}_{Rel} - \vec{v}_{Rel_{||}} \\ \vec{v}_{Rel_{||}} &= (\vec{v}_{Rel} \cdot \hat{v}_{Ecf}) \hat{v}_{Ecf} \quad (\text{projection}) \end{aligned}$$

$$\boxed{4} \quad \vec{v}_{Rel_{\perp}} = \vec{v}_{Rel} - (\vec{v}_{Rel} \cdot \hat{v}_{Ecf}) \hat{v}_{Ecf}$$

correction vector, need for unit length view vector

Summary of 5 equations found in vac.m

$$1 \quad \cos \theta = \sqrt{\hat{E}_{cf}} \cdot \frac{-\vec{\text{sat_pos}}}{|\vec{\text{sat_pos}}|}$$

$$2 \quad \lambda = |\vec{\text{sat_pos}}| \cos \theta - \left[|\vec{\text{sat_pos}}|^2 \cos^2 \theta + r_e^2 - |\vec{\text{sat_pos}}|^2 \right]^{1/2}$$

$$3 \quad \vec{v}_{\text{Rel}} = \begin{bmatrix} v_x \\ v_y \\ v_z \end{bmatrix} - \lambda (\vec{\omega} \times \sqrt{\hat{E}_{cf}})$$

$$4 \quad \vec{v}_{\text{Rel}_\perp} = \vec{v}_{\text{Rel}} - (\vec{v}_{\text{Rel}} \cdot \sqrt{\hat{E}_{cf}}) \sqrt{\hat{E}_{cf}}$$

$$5 \quad \vec{v}_{\text{EcfCorr}} = \sqrt{\hat{E}_{cf}} - \left[\frac{\vec{v}_{\text{Rel}_\perp}}{c} \right]$$

```
function vEcfCorr = vac( vEcf, satPos, satVel );
meanSiderealRotationRate = 7.292115833000E-05;
omega = [0 0 meanSiderealRotationRate]'; % earth rotational rate
speedOfLight = 299792458; % meters/second
meanEarthRadius = 6371000; % Mean Earth radius (in meters)
normSatPos = norm(satPos);
```

$$1 \quad \text{cosTheta} = \text{dot}(\text{vEcf}, -\text{satPos} / \text{normSatPos});$$

$$2 \quad \text{lambda} = \text{normSatPos} * \text{cosTheta} - (\text{normSatPos}^2 * \text{cosTheta}^2 \dots \\ + \text{meanEarthRadius}^2 - \text{normSatPos}^2)^{0.5};$$

$$3 \quad \text{satVelRelative} = \text{satVel} - \text{lambda} * \text{cross}(\text{omega}, \text{vEcf});$$

$$4 \quad \text{velPerp} = \text{satVelRelative} - (\text{satVelRelative}' * \text{vEcf}) * \text{vEcf};$$

$$5 \quad \text{vEcfCorr} = \text{vEcf} - \text{velPerp} / \text{speedOfLight};$$

$$\text{vEcfCorr} = \text{vEcfCorr} / \text{norm}(\text{vEcfCorr});$$